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# AN EXPERIMENTAL STUDY OF THE NEAR-WAKE STRUCTURE OF A CYLINDER UNDERGOING COMBINED TRANSLATIONAL AND ROTATIONAL OSCILLATORY MOTIONS 

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#### Abstract

The experimental research reported here uses particle image velocimetry to extend the study of Nazarinia et al. [1], recording detailed vorticity fields in the near-wake of a circular cylinder undergoing combined translational and rotational oscillatory motions. The focus of the present study is to examine the effect of the ratio between the translational and rotational velocities of the cylinder on the synchronization of the near-wake structures. The frequencies are fixed close to that of the natural frequency of vortex shedding. The results are presented for a fixed amplitude of rotational oscillation of 1 radian and a range of ratios between the translational and rotational velocities $\left(V_{R}\right)=[0.25,0.5,1.0,1.5]$. In particular, it was found that varying the $V_{R}$ value changed the near-wake structure. The results show that at lower values of $V_{R}=0.25$, for all of the phase differences examined, the vortices are shed in a single-row $2 S$ mode aligned in the medial plane with a slight offset from the centerline and also synchronized with the combined oscillatory motion. As $V_{R}$ increases the vortex shedding mode changes from a $2 S$ single-row to a $2 S$ double-row structure and eventually back


[^0]to the single-row (at $V_{R}=0.5$ ). Increasing $V_{R}$ further resulted in the appearance of unlocked-on regimes over the range of negative phase angles and a transition from $2 S$ to $P+S$ mode at the in-phase case There was transition back to the $2 S$ with a further decrease of $\Phi$. For a higher $V_{R}$ the range of desynchronization increased.

## NOMENCLATURE

$\theta$ Angular displacement
$\beta$ Stokes number
$\mu \quad$ Dynamic viscosity (kg.m ${ }^{-1} . \mathrm{s}^{-1}$ )
$v$ Kinematic viscosity, $\mu / \rho,\left(\mathrm{m}^{2} . \mathrm{s}^{-1}\right)$
$\omega_{\mathrm{x}}, \omega_{\mathrm{y}}, \omega_{\mathrm{z}} \quad$ Vorticity components along the $\mathrm{x}, \mathrm{y}$ and z axis respectively ( $\mathrm{s}^{-1}$ )
$\Phi$ Phase angle between rotational and translational motions
$A_{\theta} \quad$ Amplitude of rotational oscillation in radians
$A_{t} \quad$ Amplitude of translational oscillation
D Cylinder diameter
$f$ frequency of oscillation (Hz)
$f_{N}$ Natural frequency of vortex shedding (Hz)
$f_{v} \quad$ Frequency of vortex shedding $(\mathrm{Hz})$
$F_{R}$ Frequency ratio between translational and rotational oscillation $=f_{t} / f_{\theta}$
$F_{R N}$ Frequency ratio between oscillatory motion and natural frequency $=f / f_{N}$
$F_{R t} \quad$ Frequency ratio between translational oscillation and natural frequency $=f_{t} / f_{N}$
$F_{R \theta} \quad$ Frequency ratio between rotational oscillation and natural frequency $=f_{\theta} / f_{N}$
$K C$ Keulegan-Carpenter number
PIV Particle image velocimetry
Re Reynolds number based on free-stream velocity and cylinder diameter $=\left(U_{\infty} D\right) / v$
St Strouhal number
$U_{\max } \quad$ Peak velocity of an oscillating cylinder
$U_{\infty} \quad$ Free-stream velocity
$V_{R}$ Velocity ratio between translational and rotational oscillation $=U_{\text {max }_{\mathrm{t}}} / U_{\text {max }_{\theta}}$
y Translational displacement

## INTRODUCTION

Studies on flow past an oscillating cylinder can be divided into three categories depending on the motion of the cylinder: the cylinder oscillates translationally only at an angle with respect to the free-stream; the cylinder performs a rotational oscillation about its axis in mean flow; the flow is created by a circular cylinder moving with combined oscillatory translational and rotational motion in quiescent fluid or free-stream. Many researchers (such as [2-12] and many more) have shown that the vortex shedding phenomenon can be dramatically altered for the cylinder undergoing either in-line or transverse oscillation in a fluid stream. There has also been considerable research on the purely rotational oscillation case (see for example: [13-18]). The latter is a class of flow that has not received as much attention as the case of purely oscillating cylinders have, until now. Blackburn et al. [19] found that a circular cylinder undergoing combined oscillation in a quiescent fluid has the capability to generate thrust, i.e. "swimming cylinder". Nazarinia et al. [1, 20] extended their study and for the first time experimentally measured the flow around a circular cylinder undergoing combined translational and rotational oscillation in a free-stream and quiescent fluid, respectively. Placing the circular cylinder which is undergoing combined oscillatory motion in a free-stream has been shown to generate intriguing wake modes and has also been found to have the potential to reduce synchronization of the cylinder motion in the near-wake [1,21,22]. Nazarinia et al. [1] experimentally and numerically studied the flow behind a cylinder undergoing forced combined oscillatory motion. The focus of this paper was on the effect of the $\Phi$ angle between the two motions but the influence of the velocity ratio between them was not considered. They found that there is an unexpected loss of
synchronization of the wake for a finite range of phase differences. The primary focus of the reported study here is to understand and investigate the effect of $V_{R}$ on the synchronization of the cylinder motion in the near-wake. Improving our understanding of how to effect desynchronization increases the likelihood that such approaches can be used in the active or passive control of vortex-induced vibration.

## PROBLEM DEFINITION

The equations of the forcing motions, consisting of two independent oscillations: cross-stream translation and rotation, are:

$$
\begin{align*}
y(t) & =A_{t} \sin \left(2 \pi f_{t} t\right)  \tag{1}\\
\theta(t) & =A_{\theta} \sin \left(2 \pi f_{\theta} t+\Phi\right) \tag{2}
\end{align*}
$$

The flow past a cylinder undergoing combined translation and rotation oscillations depends on the six independent parameters defined by Eqns. (3)-(5) and the phase difference between the motions ( $\Phi$ ), as defined by Eqn. (2). The dimensionless quantities representative of independent variables are defined as follows:

$$
\begin{gather*}
K C_{t}=\frac{U_{\max _{t}}}{f_{t} D}=\frac{2 \pi A_{t}}{D}, \quad K C_{\theta}=\frac{U_{\max _{\theta}}}{f_{\theta} D}=\pi A_{\theta},  \tag{3}\\
F_{R t}=\frac{f_{t}}{f_{N}}, \quad F_{R \theta}=\frac{f_{\theta}}{f_{N}}  \tag{4}\\
R e=\frac{U_{\infty} D}{v} . \tag{5}
\end{gather*}
$$

The cylinder's velocity ratio, between the translational and rotational motions can also be expressed as:

$$
\begin{equation*}
V_{R}=\frac{U_{\max _{\mathrm{t}}}}{U_{\max _{\theta}}} \tag{6}
\end{equation*}
$$

Figure 1 shows a schematic of the problem studied with some relevant notations and features. The Cartesian coordinate system in use is defined such that the origin is located at the center of the circular cylinder (at $t=0$ ) with $x, y$ and $z$ representing the streamwise, transverse, and spanwise directions, respectively.

## METHODS AND TECHNIQUES <br> Experimental Setup

The experiments were conducted in the FLAIR free-surface closed-loop water channel at the Department of Mechanical and


FIGURE 1. Schematic showing the problem geometry and important parameters relevant to the combined forced oscillation and the circular cylinder model. The streamwise direction is the $x$-direction.

Aerospace Engineering, Monash University. Detailed information of the set-up can be found in [1,20]. The cylinder used was 800 mm in length with an outer diameter of 20 mm , giving an aspect ratio of 40 . The experiments were performed for a fixed average upstream velocity $U_{\infty}=0.0606 \mathrm{~m} / \mathrm{s}$ giving $R e_{\text {avg }}=1322$. The frequencies of the motions are fixed close to that of the natural frequency ( $T^{-1}=f_{t}=f_{\theta}=0.6 \mathrm{~s}^{-1} \approx f_{N}$ ). The natural frequency was found to be equal to $f_{N} \approx 0.6154 \mathrm{~s}^{-1}$. The Strouhal number based on this frequency is about $S t \approx f_{N} D / U_{\infty}=0.203$ and the Strouhal number of the forcing is $S t_{t} \approx f_{t} D / U_{\infty}=0.198$. The results of experiments presented here are for (a) a fixed amplitude of rotational oscillation of 1 radian and a range of $0.25 \leq V_{R} \leq 1.5$ and (b) Case 5 has a reduced amplitude of rotational oscillation of 0.5 radians and $V_{R}=1$. For each $V_{R}$ case the phase difference angle was varied during measurements. Table 1 summarizes the data points measured in the present study.

The cylinder was oscillated translationally and rotationally by using two high-resolution stepper motors. The translational stepper motor actuated the rodless in-line mounting actuator and the rotational stepper motor was connected directly to the vertically mounted cylinder. The stepper motors were controlled using a two-axis indexer and two high-resolution drivers (running at 50800 steps rev $^{-1}$ ). A pure sinusoidal profile, as defined in Eqns. 1 and 2, is used throughout the paper. A TTL-signal triggered other devices (camera and laser), thus images could be captured at pre-selected phase angles in the oscillation cycle (phase-locked).

## Experimental Techniques

The method used here to characterize the wake of this forced cylinder is via PIV. The PIV set-up was based on that originally described by Adrian [23] and developed in-house over the past decade [24]. The flow was seeded with spherical granular polyamide particles having a mean diameter of $55 \mu \mathrm{~m}$ and specific gravity of 1.016 . The particles were illuminated using two mini-YAG laser sources (Continuum Minilite II Q-Switched).

TABLE 1. SUMMARY OF THE VALUES OF $A_{t}$ AND $A_{\theta}$ USED WHEN $F_{R}=1.0$.

| Case | $A_{t}$ | $A_{\theta}(\mathrm{rad})$ | $V_{R}$ | $U_{\mathrm{max}_{\mathrm{t}}}\left(\mathrm{ms}^{-1}\right)$ | $U_{\text {max }_{\mathrm{t}} / U_{\infty}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | $D / 8$ | 1.0 | 0.25 | 0.0094 | 0.16 |
| 2 | $D / 4$ | 1.0 | 0.5 | 0.0188 | 0.31 |
| 3 | $D / 2$ | 1.0 | 1.0 | 0.0377 | 0.62 |
| 4 | $3 D / 4$ | 1.0 | 1.5 | 0.0565 | 0.93 |
| 5 | $D / 4$ | 0.5 | 1.0 | 0.0188 | 0.31 |

The plane of interest for these experiments was orthogonal to the cylinder's axis (xy-plane) and downstream ( $x$-direction) of the cylinder. A small section of the cylinder is replaced by a thin-walled transparent cylinder, whose interior is filled with distilled water. It is located at about $9 D$ from the end of cylinder. The measured $x y$-plane is located through the centre of this window. The thickness of the laser sheet was measured to be approximately less than 2 mm . Pairs of images were captured on a high resolution CCD camera with a maximum resolution of $4008 \times 2672$ pixels. The camera was equipped with a 105 mm lens (Nikon Corporation, Japan). At a particular phase of the oscillation cycle, a number of image pairs over successive cycles were taken and stored for further processing. The timing of the laser and camera triggering was controlled by a special in-house designed timing unit, with an estimated accuracy of $1 \mu \mathrm{~s}$.

Each image pair was processed using in-house PIV software [24]. This software uses a double-frame, cross-correlation multi-window algorithm to extract a grid of displacement vectors from the PIV images. An interrogation window of $32 \times 32$ (with an initial window size of $64 \times 64$ ) pixels was found to give satisfactory results with $50 \%$ overlap. More than $98 \%$ of the vectors were found to be valid for all the experiments. It was possible to obtain a measurement resolution of $127 \times 127$ vectors in each field of view. The overall field of view was $4008 \times 2672$ pixels ( $6.0 D \times 6.0 D$ ).

Our PIV set-up and technique have been validated against the previous numerical and experimental results $[25,26]$ for the $x y$ - and $y z$-plane measurements. The validation case studied is for a purely translational oscillation in a quiescent fluid. Further information can be found in [20].

## RESULTS AND DISCUSSION

In this section the results are presented for velocity ratios of $0.25,0.5,1.0$ and 1.5. The wake profiles around the cylinder in the streamwise direction $x y$-plane for $F_{R}=1.0$ and $R e_{\text {avg }}=1,322$ are investigated. Due to the change in the temperature of the water the actual $R e$ number of each set of experiments might be
marginally different from the averaged value, which is presented for each set of experiments in the relevant figure. It should also be noted that the field-of-view does not allow us to see the flow structures that occur farther downstream, however the near-wake flow structures are the main focus of this study. For every value of $V_{R}$ studied, the effect of a change of phase difference angle on the synchronization of the near-wake has also been examined. However, in this paper only selected phases are discussed and presented.

Figure 2 presents the near-wake motion phased-locked vorticity and root-mean-square (rms) vorticity contours taken at $t=T$ for various $V_{R}$ values at $\Phi=0^{\circ}$ i.e. in-phase. The image at the top left of Fig. 2 shows the case where the velocity ratio is 0.25 . In this case we observe a $2 S$ mode ( 2 single vortices of opposite sign shed per period) in a single-row aligned in the medial plane but with a slight offset from the centreline. The field-of-view does not allow us to see the double-row that should occur further downstream (see for example [27-29]). The classification of the different vortex modes is given by [9]. Throughout the observations it can be seen that the near-wake vortices remain coherent in the near-wake and are synchronized with the translational motion, at least up to $6 D$ downstream. This synchronization is characterized by the repeatable pattern of the vortex shedding. Examination of the $U / U_{\infty}$ and $V / U_{\infty}$ velocity contours, not shown here, also suggests that the velocity structures are not changed significantly by changing the phase difference angle and that the vortices are synchronized with the translational motion, i.e. locked. Note that setting the $V_{R}$ to such a small value means the rotational oscillatory speed is much faster than the translational. This locking-on can be explained by the rotational oscillation adding momentum to the flow.

Figure 2 b shows the phased-locked vorticity and rms vorticity contours for $V_{R}=0.5$. In this case the vortices are still shed in a $2 S$ single-row mode and are coherent. However,the vortices are arranged closer to each other and are less well-aligned with the medial plane, suggesting an earlier double-row transition. The rms vorticity, image (b) of Fig. 2, along with $U / U_{\infty}$ and $V / U_{\infty}$ velocity contours not shown here, have a similar trend and configuration to the previously discussed case, confirming that the near-wake still synchronizes with the combined motion. However, a change in the phase difference between the two motions results in a change of shedding mode from a single-row to double-row.

Increasing the $V_{R}$ further to 1.0, image (c) of Fig. 2, changes the vortex shedding mode to the signature of a $P+S$ mode (a single vortex and a vortex pair formed per cycle) in the near-wake, in comparison with the $2 S$ mode observed in the previous cases. For this case, the vortices are shed widely apart (nearly $4 D$ ), which is readily explained by the rotational oscillation adding momentum to the translational motion. The resulting strain favours a transition to the $P+S$ wake [30].

Figure 2 d shows the phased-locked vorticity and rms vor-


FIGURE 2. Motion phased-locked vorticity contours (lines) and root-mean-square vorticity (gray-scale) contours taken at the motion-phase of $t=T$ for $A_{\theta}=1.0, \Phi=0^{\circ}, f_{N}=0.6 \mathrm{~Hz}, F_{R}=1$ and $R e_{\text {avg }}=1322$. The near-wake vorticity is shown for different $V_{R}$ values. The flow direction is from left to right. Root-mean-square vorticity contours are evenly spaced over the range $[0.02: 0.1]$; with $\Delta \omega_{z}=0.02$, and vorticity contours are evenly spaced over the range $[-0.1: 0.1]$; with $\Delta \omega_{z} r m s=0.01$.
ticity contours for $V_{R}=1.5$. The shedding mode for this case is $P+S$. The vortices are more defined in the vortex street and are observed to be synchronized with the motions. The vortices are more widely aligned than those seen in image (c).

In the results presented to this point the change of $V_{R}$ was investigated by varying $A_{t}$ and keeping $A_{\theta}=1$ radian. In Fig. $3 A_{\theta}$ was reduced to 0.5 radians while $V_{R}$ was set to 1.0 . This allows us to directly compare the vorticity patterns with those of image (c) of Fig. 2, which has the same $V_{R}$ but $A_{\theta}=1$ radian. It can be observed that the vortices in Fig. 3 are all larger in size than those shown in Fig. 2 and are well-formed. The near-wakes for all of the $\Phi$ values are synchronized with the motions and the shedding mode is $2 S$. It can be seen that even though the $V_{R}=1.0$ is the same for the two cases 3 and 5 of Table 1 the near-wake structure is completely different. No transition between shedding modes can be observed as $\Phi$ is varied. In case 3 , because the rotational amplitude is greater than that of case 5 , it adds momentum more compactly into the flow, hence the vortices are more compact and the phase difference does not influence the synchronization of the near-wake nor the vortex shedding mode. Root-mean-square vorticity and velocity contours all confirm the synchronization of the near-wake. The velocity patterns are qualitatively similar to


FIGURE 3. Motion phased-locked vorticity contours taken at the motion-phase of $t=T$ for $f_{N}=0.6 \mathrm{~Hz}, F_{R}=1, A_{t}=D / 4, A_{\theta}=0.5 \mathrm{ra-}$ dian and $R e_{\text {avg }}=1322$. The near-wake vorticity is shown for different $\Phi$ values. The flow direction is from left to right. Vorticity contours are evenly spaced over the range $[-0.1: 0.1]$; with $\Delta \omega_{z} r m s=0.01$.
the locked previous cases and no evidence of transition between states can be observed.

Figures 4 and 5 present the near-wake motion phased-locked vorticity and rms vorticity contours and $V / U_{\infty}$ velocity contours, respectively, taken at $t=T$ for various $V_{R}$ values at $A_{\theta}$ and $\Phi=-90^{\circ}$. The near-wake vortex structures and vortex modes of images (a) and (b) in Fig. 4 are similar to those of Fig. 2. The vortices are not separated widely and remain less than $1 D$ from the centreline. The $2 S$ mode in a single-row aligned in the medial plane with a slight offset from the centreline was observed. Increasing the $V_{R}$ value at the same $\Phi$ was found to change the synchronization effect dramatically as can be seen in Fig. 4c and d. The rms vorticity contours clearly show how the near-wake has changed its structure. Contrary to the other cases shown in images (a) and (b), these two cases were not synchronized with the motions beyond $2 D$ downstream. The effect of this loss of synchronization can be seen in the rapid downstream dissipation of the mean vortex structures. Only the two vortices near the cylinder remain coherent. This a priori surprising phenomenon might be explained by the fact that the separation between the two rows of vortices is smaller and that this arrangement of vortices is not stable. Similar behaviour has been found behind elliptical cylinders [29].

Comparing the $V / U_{\infty}$ and rms vorticity contours of this case


FIGURE 4. Motion phased-locked vorticity contours (lines) and root-mean-square vorticity (gray-scale) contours taken at the motion-phase of $t=T$ for $A_{\theta}=1.0, \Phi=-90^{\circ}, f_{N}=0.6 \mathrm{~Hz}, F_{R}=1$ and $R e_{\text {avg }}=1322$. The near-wake vorticity is shown for different $V_{R}$ values. Of particular interest is the unlocked (asynchronous) wake with the imposed translational motion for the velocity ratios of 1.0 and 1.5 . The flow direction is from left to right. Root-mean-square vorticity contours are evenly spaced over the range [0.02:0.1]; with $\Delta \omega_{z}=0.02$, and vorticity contours are evenly spaced over the range $[-0.1: 0.1]$; with $\Delta \omega_{z} r m s=0.01$.
(Fig. 5 and 4) with the previous cases shows that there is a dramatic change of patterns in the velocity contours between the synchronized and desynchronized cases. The synchronized velocity patterns are all well organised and qualitatively similar to each other. Figure 5 clearly shows that the $V$ velocity contour pattern has changed dramatically for the unlocked cases and for distances farther downstream beyond approximately $3 D$ the dissipation of vortices is more apparent.

For the $V_{R}=1$ case shown in FIg. 4 the contours of vorticity that have been phase-averaged with the cylinder's motion appear to coincide with those of the rms vorticity patterns, despite the wake being unlocked for this case. Similar trends were also found to apply in the cases of which is not shown here is applicable to the $\Phi= \pm 180^{\circ},-120^{\circ}$ and $-150^{\circ}$. They are not well formed and the structures of the vortices are not coherent. This pattern shows that the flow is not synchronized and the perturbations are clearly visible in those regions, hence they are all unlocked.


FIGURE 5. Motion phased-locked $V / U_{\infty}$ velocity contours taken at the motion-phase of $t=T$ for $A_{\theta}=1.0, \Phi=-90^{\circ}, f_{N}=0.6 \mathrm{~Hz}, F_{R}=1$ and $R e_{a v g}=1322$. The velocity contours are shown for different $V_{R}$ values. Of particular interest is the unlocked (asynchronous) wake with the imposed translational motion for the velocity ratios of 1.0 and 1.5. The flow direction is from left to right. $V / U_{\infty}$ velocity contours are evenly spaced over the range $[-0.2: 2.4]$; with $\Delta\left(V / U_{\infty}\right)=0.2$.

## CONCLUSIONS

In this work we have experimentally investigated the nearwake flow structures of flow past a cylinder undergoing a combined translation and rotation oscillatory motion. The effect of the velocity ratio between the two forced motions for a given Reynolds number and combination of $\Phi$ reveals that regular vortex shedding can be suppressed for particular phase differences and $V_{R}$ values. The range of phase difference for which the suppression occurs depends on the $V_{R}$ and $F_{R}$ of the oscillations and the oscillation frequency ratio to natural vortex shedding frequency of a fixed cylinder. At the lower values of $V_{R}$, over the range of phase difference angles studied, no transition to different wake modes is observed. All the vortices in the near-wake region of the cylinder are synchronized and coherently shed in the $2 S$ mode. Increasing the $V_{R}$ value beyond a certain value changes the wake modes and the synchronisation of the vortices in the near-wake. The vortices become unlocked and less coherent as $V_{R}$ is increased further for a range of $\Phi$. The effect of changing translational and rotational amplitudes for a given $V_{R}$, i.e. $V_{R}=1.0$, was also investigated. It appears that at higher $A_{t}$ and lower $A_{\theta}$ the vortices in the near-wake are more compact and larger in size. Changing the $A_{t}$ and $A_{\theta}$ did not, however, seem to
have an effect on the synchronisation of the vortices and the wake shedding modes. This experimental study raised several interesting features which determine the wake modes in the near-wake of the combined oscillatory cylinder in free-stream. One particularly interesting question is whether the desynchronized state has a chaotic or quasi-periodic form. This will require further investigation using higher frequency sampling than is possible with the current PIV system. The answer is likely to be revealed by a combination of Laser Doppler Anemometer experiments and numerical modeling.

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